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## **AUTHORITY**

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# ARMAMENT DESIGN ESTABLISHMENT MINISTRY OF SUPPLY

THE DETERMINATION OF THE OPTIMUM SIZE OF A.P.C.B.C./D.S.
SHOT AND OF THE TUNGSTEN CORE OF A.P./D.S. PROJECTILE
FOR MAXIMUM PERFORATION OF ARMOURED PLATE

F. BOOKER

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OFFICIAL SECRETS ACTS

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THE DETERMINATION OF THE OPTION SIZE OF A.P.C.P.C./D.S. SHOT AND ESTIMATE TURGETTEE CORRECT A.P./D.S. PRAIRITILE ATT MAKING PERFORATION OF ALEXONS PERFORATION OF

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Recommended for publication

T. Poberts, D.2.

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Approved for publication

Abstract

In this paper the optimum sizes of A.P.C.B. L/D.S. shot and A.P./D.S. Tungsion cores for perforating armour plate are determined.

It is shown that the values are given approximately by the ratios:-

Shot calibre/# 11 calibre
Tomgaten cors : slibre/full calibre

but more monumate values, which depend upon the ballisticweights and the range, can be determined. It is concluded also that slight reduction in these ratios may be adopted if measures without seriously reducing the thickness of plate perforated.

Amessout Design Establishment, Ministry of Supply, Fort Salstoni, Kent,

Phone: Severpaks 5211

April, 1952

#### THE DETERMINATION OF THE OFTIMUM SIZE OF A.P.C.B.C./D.S. SHOT AND OF

#### THE TUNGSHE CORE OF A.P./D.S. PROJECTILE FOR MAXIMUM

#### PER RATION OF ARMOUR PLATE

#### Introduction.

Discarding is not Proporties on the main, a shot of calibratement than the gum and fitted with a light alloy body (the discarding salest) whose calibre suits the gum hore and centres the shot.

By this construction thigher velocity is given to the shot than would be obtained if fixed from a punt of calibre equal to the shot; and the velocity at the target would be expressionally higher for perforation.

Perforation of the target is dependent upon the weight, disnessed and velocity of the shot and as it is possible to design a D.S. projectile with varying sizes of shot is required to know the best size which will effect maximum perforation. The plate.

#### Assumptions.

The assumpt: or a made in this paper are those also given in A.R.E. Weapons Branch is no No. 1/5:

(a) Internal ballistic relation:

(Projectile weight plus half the charge) times the square of the muzzle velocity as constant for the same charge at all velocity.

di

- (b) Shatter does not a our i. any case
- (c) The projectiles are stable
- (d) The weight of the liscarding components of the projectiles is constant for the same type of projectile
- (e) log<sub>10</sub> C used in the De Marie formula, for a given angle of impact; is constant
- #) The musice velocities considered cover 3700-6000 ft/sec.

o stated hat

- (a) is a near enough approximation
- (b) is an anticipation
- (c) can be made true (d) A.R.B. and A.D.B. and from experience that this is reasonably the
- (e) this has not hear convincingly disproved by experiments

#### - Velocity Equation

The normal metho: for determining the velocity of shot at various ranges is that due to Siscoi.

This method does to lend itself to being introduced into the problem and, in the following, a sime . equation relating range and velocity is obtained,

One form of the issistance equation is  $\frac{1}{2}\rho C_{D} \nabla^{a} A(\frac{a}{a})$ , where "a" is the velocity of sound in air.

The Text Book of the istics and Gunnary 1946 also states that a reasonable fit for valocities be see: 3700 and 6000 ft/sec is given by

Resistance . IV

Hence we can wai a resistance in the form

where K and K, are ballist on stants and the equation of motion may be written

which leads to 
$$x = \frac{\pi}{K_1} \cdot \frac{\pi}{8} g d \cdot \frac{1}{8} (\nabla_1^4 - \nabla_x^4)$$
 (1)

WINETE

x = range i ft.

Van velocit, at r. ge x.

d = dismete of bey in ft.

W = weight b.d. in 1b.

Table I gives the values of  $K_1(K\sigma)$  determined from the results obtained by the Siacci method for for subminojectiles at the ranges 500, 1000 and 2000 yds.

TABLE I

Sub Projectile	¥ 1b.	d ins.	I feet	2/8	N	K <sub>1</sub> (ko)	Mean K 1(K d)	K <sub>a</sub> (Ko)	Mean K <sub>2</sub> (Ko)
A B C D	6.78 11.8 14.32 21.82	2,08 2,515 2,675 3,062	1500 1500 1500 1500	5230 4700 4400 3910	5032 4536 4249 3783	.1676 .1687 .1680 .1690	.1685	.9251 .9122 .9009 .8812	.9048
A 8 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	6.78 11.8 14.32 21.82	2.675 2.675 3.062	5000 3000 3000 3000	5230 4730 4400 3910	4835 4373 4300 3658	.1678 .1688 .1683 .1683	.1683	.9228 .9094 .8949 .8742	<b>.</b> 9003
A B C D	6.78 11.8 14.32 21.82	2.08 2.51' 2.675 3.062	6000 6000 6000 6000	5230 4700 4400 3910	4446 4050 3804 3408	.1680 .1690 .1684 .1688	.1686	.9158 .9039 .4889 .8707	.8948
				* **			.1685		.9

X . 11.06 (7. -7.

bero d is in inches x is in yards.

(2)

A simpler form may be produced by assuming the resistance varies as the velocity thus

Resistance # K2(Ko)daV

$$x = \frac{W}{X_{2}(X^{0})gd^{2}} \xrightarrow{11/4} (V_{1} - V_{X}) \qquad \text{where } x = feet \\ d = ins$$
 (3)

The values of  $K_2(K\sigma)$  are also given in Table I showing, they are reasonably constant. Using the mean value we have

$$\Sigma = \frac{1.656W}{dz} (\nabla_z - \nabla_z) \tag{4}$$

where X is in yards
W is in lb
d is in ins.
W is in ft/sec.

The projectiles considered in this paper have the same  $(K\sigma)_1$  value of 1.1 and therefore equation (4) will be used throughout. If, however, the form of the shot is changed, giving a new value  $(K\sigma)_2$  the range X as determined by equation (4) is changed to

#### The A.P.C.B.C./D.S. Projectile - Optimum Sise.

#### 1. Based on the Full Calibre

In this assessment the weight of the Balliatic cap is included in the weight of the body considered as perforating the target. This may not be strictly true but as its weight is a small fraction of the shot weight the error resulting is probably small. A typical form of the projectile is shown in Fig. 1.

Let

Dismeter of Full calibre shot  $= d_1$ Dismeter of sub calibre shot  $= d_2$ Weight of Full calibre shot  $= W_1 = kd_1^2$ Weight of sub calibre shot  $= W_2 = kd_2^2$ Weight of discard  $= W_3 = kd_2^2$ Weight of projectile  $= W_3 + W_3$ Weight of charge = C (constant)
Waxale Velocity of F.C. shot  $= V_1$ Mussle Velocity of sub shot  $= V_2$ 

Then for constant energies as implied in assumption (s)

$$(W_{1} + \frac{C}{2}) V_{1}^{2} = (W_{2} + W_{2} + \frac{C}{2}) V_{3}^{2}$$

$$V_{9} = V_{1} \frac{1 + \frac{C}{2W_{1}}}{K^{2} + \frac{C}{2W_{2}} + \frac{W}{W_{1}}}$$
where  $K = d_{0}/d_{1}$  (5)

We also have, Equation(4), I =  $1.656\frac{W}{d}$  ( $V_1 - V_2$ )

and Milnes perforation equation

$$\frac{\pi \gamma^2 \cos^2 \theta}{d} = C_1(\frac{4}{d})^{1.45} \tag{6}$$

Applying these equations to the Fell calibre and sub calibre shot

Hence the ratio  $R = \frac{t_0}{t_0}$  is

$$R = \frac{t}{t} \frac{a_1}{a_2} \cdot \frac{O_1}{O_2} e^{\frac{t}{1-h}} \cdot \frac{1}{f} \left[ \frac{1}{V_1} \frac{1}{a_1} \cdot \frac{J^2 A B}{(X^2 b)} \frac{1}{f} - \frac{h}{K^{\frac{3}{2} - 4B}} \right]^{\frac{B}{B^2 4 B}}$$

To find the options R, dR = 0 which from (7) gives

$$.725 - \frac{3}{2} \frac{X^{4}}{(X^{2}+b)} + 1 \frac{(X^{2}+b)^{\frac{1}{2}}}{X} = 0$$

where  $a = 1 + \frac{\sigma}{2\overline{\pi}_1}$   $b = \frac{\sigma}{2\overline{\pi}_1} + \frac{\overline{\pi}_2}{\overline{\pi}_1} \qquad (7)$   $K = \frac{d_2}{d_4}$ 

$$h = \frac{xd_i^3}{1.656} w_i$$

$$f = [v_i - h]^{\frac{3}{4}\frac{3}{4}}$$

$$t = \frac{285h}{v_i a \frac{3}{4}}$$

(8)

It will be seen later that the effect of the third term is small, though not entirely negligible, and neglecting this, a first approximation from (8) gives

$$\mathbf{x}_{\mathbf{b}} = .969 \quad \mathbf{\bar{b}}$$
 (9)

Variation of the ration  $\frac{t_2}{t_1}$  with the Range X

For any ratio  $K = \frac{d_2}{d_1}$  the ratio of  $\frac{t_2}{t_1}$  may be determined from (7) which reduces to

$$\frac{t_1}{t_1} = \left(\frac{o_1}{o_2}\right)^{\frac{1}{1-2}} \frac{1}{k} \cdot \cos \left[1 - \frac{1 - \left(\frac{a_1}{1-a_2}\right)^{\frac{1}{2}}}{1 - \frac{1}{1-a_2} \cdot \frac{d_2}{1-a_2}}\right]^{\frac{1}{1-a_2}}$$
(10)

These equations will now be applied to current problems.

#### Example 1

20 Pr. A.P.C.B.C./D.S. - Design D2(L)6943

Dismeter of gum % 3.3 ins = d<sub>1</sub>
Dismeter of 20 Pr. sub-projectile # 2.08 ins
E.V. of 20 Pr. # 5020 ft/sec.
Charge weight # 9.6 lb. # C
Weight of sub-shot and cusp # 5.2 lb. # W<sub>2</sub>
Weight of the discard # 2.03 lb # W<sub>3</sub>

These particulars apply to the existing design which may be the optimum sine. The following will show what the optimum sine is.

The weight of a Full Galibre shot =  $5.2(\frac{3.3}{2.08})^8$  = 20.8 lb = W<sub>1</sub> Equating Energies,

$$(5.2 + 2.03 + \frac{9.6}{2})$$
 5020° =  $(20.8 + \frac{9.6}{2})$ V<sub>1</sub>°

Velocity of F.O. shot =  $V_1 = 3440$  ft/mec.

almo

$$b = \frac{9.6}{2 \times 20.8} + \frac{2.03}{20.8} = .2308 + .0976 = .3284$$

From equation (9) the optimum value of E to first approximation is

$$K_0 = .969$$
  $\sqrt[4]{.3284} = 0.668$ 

or the optimum size of sub-shot =  $.668 \times 3.3 = 2.2$  ins and not 2.08 ins as in design D2(L)6943.

For a more accurate evaluation of  $K_0$  equation (8) is used and commencing with a value of the order of  $K_0$  calculated above the following tabulated method is applied. An alternative direct solution is also given in the appendix.

K .67 .68 .69 .7 .71  
K .3008 .3144 .3285 .343 .3579  
K .6292 .6428 .6569 .6713 .6863  

$$\sqrt{K}$$
 .7932 .8018 .8105 .8193 .8284  
 $\sqrt{K}$  .4780 .4891 .5 .5109 .5215  
 $\sqrt{K}$  .1184 1.175 1.175 1.17 1.167 So for

Eq. (3)  $(x = 500) + .012 - .005$ 
 $(x = 2000) + .054 + .037 + .02 + .004 - .008$ 

Also  $i = \frac{.285}{.} \frac{x}{1.656} \frac{x}{1.656}$ 
 $= \frac{.285 \times 3.3^{2} \times 3.3^{2} \times 3.40 \times 3.23 \times$ 

So the optimum for range of 500 yds is  $.677 \times 3.3 = 2.23$  ins for range of 2000 yds is  $.703 \times 3.3 = 2.32$  ins

It is seen, therefore, that the optimum size depends upon the range but as one size of shot only can be adopted in practice, a compromise for the above is a dismeter of 2.27 is si.e. K = .688.

From equation (10) the ratio of t<sub>2</sub>/t<sub>1</sub> may now be calculated for this mean optimum size, and for the ranges 500 and 2000 yds.

$$\frac{\xi_{g}}{\xi_{i}} = \left(\frac{Q_{i}}{C_{g}}\right)^{\frac{1}{2+3}} \frac{1}{.688} \cdot \text{and} \left[1 - \frac{1 - \left(\frac{1.2308x.688^{3}}{.688^{3} + .3284}\right)^{\frac{1}{2}}}{1 - X \frac{3.3^{2}}{1.656 \times 20.8x3440}}\right]^{\frac{1}{2+3}}$$

$$= \left(\frac{Q_{i}}{C_{g}}\right)^{\frac{1}{2+3}} 1.16 \left[1 - \frac{.0662}{1 - .00009x}\right]^{\frac{1}{2+3}}$$
For range X = 500 yds
$$\frac{\xi_{g}}{\xi_{i}} = 1.05 \left(\frac{Q_{i}}{C_{g}}\right)^{\frac{1}{2+3}}$$

$$X = 2000 \text{ yds} \qquad \frac{\xi_{g}}{\xi_{i}} = 1.035 \left(\frac{Q_{i}}{C_{g}}\right)^{\frac{1}{2+3}}$$

So at 500 yds the sub-shot is 5% better than the F.C. projectile and at 2000 yds the sub-shot is 34% better than the F.C. projectile assuming assumption (e) applies i.e.  $c_1 = c_2$ 

#### Example 2

#### 105 m. A.P.C.B.C./D.S. - D2(L) 6816

Dissector of gum = 105 mm. = 4.134 ins.
This shot is a scale-up of the 20 Pr. sub-shot 2.08 ins. dismeter and weight 5.2 lb.
Therge weight = 16.5 lb.
Weight of discard = 4.22 lb.

Weight of F.C. shot =  $5.2(\frac{4.134}{2.08})^8$  = 41.1 lb =  $W_A$ 

For the estimation of the M.V. of the F.C. shot the only information available for a sub shot, and the charge used above (16.5 lb) is a sub-shot and discard weight of 19 lb having an M.V. of 4400 ft/sec.

Based on this, the velocity of the F.C. shot is given by

$$(19 + \frac{16.5}{2}) 4400^{\circ} = (41.1 + \frac{16.5}{2}) V_{i}^{\circ}$$
  
V. = 3270 ft/sec.

bow

$$b = \frac{16.5}{2241.1} + \frac{4.22}{41.1} = .2 + 103 = .503$$

And the optimum mime to the first approximation is

or the optimum size =  $.651 \times 4.134 = 2.69$  ins.

again applying the tabular method to obtain more accurate values

K .65 .66 .67 .68 
$$\ell = \frac{285}{3270x1.28} \cdot \frac{1.131.8}{1.656x41.1} = .00002$$

K\* .2746 .2875 .3001 .3144

K\*+b .5776 .5905 .6038 .6174

 $\sqrt{x^6+b}$  .76 .7684 .777 .7858

K\*/K\*+b .4754 .4869 .498 .5093

 $\sqrt{K^6+b}/K$  1.169 1.164 1.16 1.155

Eq. (8)  $X = 2000 + .013 = .004$ 
 $X = 2000 + .046 + .03 + .014 = .005$ 

So the optimum size for the range of 500 yards = .658 x 4.134 = 2.72 ins for the range of 2000 yds = .678 x 4.134 = 2.81 ins or a mean of 2.76 ins or K = .67 and the ratio

$$\frac{t_{a}}{t_{1}} = \left(\frac{o_{1}}{o_{2}}\right)^{\frac{1}{2-4}} \frac{1}{.67} = \frac{1}{.67} \left[1 - \frac{1 - \left(\frac{1.2x.67^{2}}{.67^{2} + \frac{1}{203}}\right)^{\frac{1}{2}}}{1 - \frac{x}{1.656xi.1.x3270}}\right]^{\frac{1}{2}}$$

$$= \left(\frac{o_{1}}{o_{2}}\right)^{\frac{1}{2-4}} = 1.17 \left[1 - \frac{.97}{1 - .000076x}\right]^{\frac{1}{2-4}}$$

For X = 500 yds 
$$\frac{t_2}{t_1} = 1.05 \left(\frac{a_1}{0.9}\right)^{\frac{1}{4-4.5}}$$
  
X = 2000 yds  $\frac{t_2}{t_1} = 1.057 \left(\frac{a_1}{0.9}\right)^{\frac{1}{4-4.5}}$ 

So at 500 yds the sub-shot is 5% better than the F.C. projectile and at 2000 yds the sub-shot is 32% better than the F.C. projectile.

#### Pirst Summery.

1. Equation (5) shows, to a Mirst approximation, the optimum size relative to the F.C. size is given by

whence b depends upon the charge weight, the discard weight and weight of the F.C. shot. A closer value is obtained by use of equation (8) but the examples show more clearly that the optimum mise varies as the range varies.

- The examples also show that a rough rule for the optimum size is 2/3
  of F.C. size.
- 3. Equation (10) gives the ratio of plate thickness capable of being perforated by the sub-shot and F.C., and shows that as the range increases this ratio decreases slightly.
- 4. Equations (7) and (8) show that the optimum size is neither affected by the angle of attack 0 when this is constant for both the subshot and F.C. shot, nor by any difference in the logic C.

Differences in the values of log C would however affect the perforation ratio  $t_2/t_1$ , as shown by equation (C) if such differences existed. S.A.B. however consider that where similar shape shot are used  $\log_{10} C$  is reasonably constant and that the relationship between  $\log_{10} C$  and t/d given in Proc. 26399 –  $\log_{10} C$  = a + b $(\frac{t}{d}-1)$  – is doubtful for general use.

5. In order to reduce any argument as to the effect of possible differences between  $\log_{10} C_p$  and the weights of discards (noting that in fact the F.C. shot does not carry a discard) the following examples approaches the problem as the designer would do. That is he first estimates the weights and obtains the ballistics for a size of shot of the order of the optimum size and then with this data proceeds to determine the optimum size. By so doing the  $\log_{10} C$  value of discard weight is less liable to vary with slight differences in shot size.

#### Alternative Method based upon Designers Preliminary estimates,

In the following, suffix 1 applies to the size of the designers first assessment and suffix 2 to the optimum size shot.

	Tentative Design of Sub-cal Projectile	Required Optimum Sime
Weight of sus-shot Disoard weight Charge weight M.V.	= V, = V, = V,	= W <sub>a</sub> = kd <sub>a</sub> * = W <sub>a</sub> = C = Y <sub>a</sub>

It is again the ration  $\frac{d_0}{d_1}$  but related to the tentative design.

Equating Energies 
$$\begin{pmatrix} \Psi_1 + \Psi_0 + \frac{Q}{2} \end{pmatrix} \nabla_x^2 = \left(\Psi_X + \Psi_0 + \frac{Q}{2}\right) \nabla_y^2$$
 
$$\Psi_y = \Psi_1 \left( \frac{\Psi_1 + \Psi_0 + \frac{Q}{2}}{\Psi_2 + \Psi_0 + \frac{Q}{2}} \right)^{\frac{1}{2}} = \Psi_x \left( \frac{1}{K^2 + b} \right)^{\frac{1}{2}}$$

From (4) 
$$X = 1.656 \frac{\Psi_{1}}{d_{1}} (\Psi_{1} - \Psi_{1X}) \qquad I = 1.656 \frac{\Psi_{0}}{d_{2}} (\Psi_{2} - \Psi_{2X})$$

$$\Psi_{1X} = \Psi_{1} - \frac{d_{1}^{2} X}{1.656 \Psi_{1}} \qquad \qquad \Psi_{2X} = \Psi_{2} - \frac{d_{2}^{2} X}{1.656 \Psi_{2}}$$

$$= \Psi_{2} - \frac{X^{2}}{1.656 \Psi_{1}} \cdot \frac{1}{X}$$
From (6) 
$$t_{1} = \frac{2^{4} \sqrt{\frac{2}{4} \sqrt{\frac{2}{1} \times \cos^{2} \theta}}}{\sqrt{\frac{2}{c_{1} d_{1}^{2} \times \Psi_{2}^{2}}} \cdot \frac{1}{X^{2}}$$

and the ratio R reduces to

$$R = \frac{t_a}{t_1} = \left(\frac{c_1}{c_2}\right)^{\frac{1}{1/2}\frac{1}{2}} \frac{1}{2} \left( \sqrt{a^2 \frac{1}{K^2 + b}} - \frac{1}{K^{-1/2}} \right)^{\frac{1}{1/2}\frac{1}{2}} \tag{11}$$

which is the same form as (10) but the values are now

$$h = \frac{X}{1.656} \frac{d_1^{\frac{1}{2}}}{1.656} \frac{d_2^{\frac{1}{2}}}{1.656} \frac{d_2^{\frac{1}{$$

Again the optimum sine is given by
$$\frac{715 - \frac{1}{2} \frac{R}{R + b} + 1}{R} + \frac{(R^0 + b)^{\frac{1}{R}}}{R} = 0$$
(12)

and to a first approximation

and the variation of to with range X by

$$\frac{t_{3}}{t_{1}^{2}} = \left(\frac{c_{1}}{c_{3}}\right)^{\frac{1}{2} + \frac{1}{2}} = \frac{1 - \left(\frac{1}{2} + \frac{1}{2}\right)^{\frac{1}{2}}}{1 - \frac{1}{2} + \frac{1}{2} + \frac{1}{2}} = \frac{1}{2} + \frac{1}{2} = \frac{1}{2} + \frac{1}{2} = \frac{1}{2$$

Equation (12, (13, (14) are similar to (4, (3, (10) but now related to the Designers tentative size estimate.

This alternative method is now applied to the problems of examples 1 and 2.

#### Examples 5

This is the problem of example 1 but now

Tentative shot dimester = 2.08 ins = d<sub>1</sub>
Weight of chot and cap = 5.2 lb = W<sub>1</sub>
Weight of discard = 2.05 lb = W<sub>2</sub>
Weight of charge = 9.6 lb = C
N.V. of shot = 5020 ft/sec. = V<sub>2</sub>

For a first approximation, equation (13 gives

$$b = \frac{9.6}{253.2} + \frac{2.03}{5.2} = .923 + .3885 = 1.3315$$

$$a = 1 + b = 2.3315$$

now using (13) to obtain more accurate values for ranges of 500 and 2000 yds

So the optimum size for range of 500 yds = 1.075 x 2.08 = 2.24 ins. for range of 2000 yds = 1.11/5 x 2.08 = 2.32 ins.

These values are in good agreement with example 1.

The practical optimum is 2.27 ins as before but I now related to the tentative design size is 1.091.

From 11.

$$\frac{t_2}{t_1} = \left(\frac{o_1}{c_9}\right)^{\frac{1}{148}} \quad \frac{1}{1.091}, \text{asso} \left[1 - \frac{1 - \left(\frac{c_9}{1.0915 + c_9} \cdot 091^9\right)^{\frac{1}{2}}}{1 - \frac{X}{2.08}} \right]^{\frac{1}{1656 \times 5}} \\
= \left(\frac{o_1}{c_9}\right)^{\frac{1}{148}} \quad .966 \left[1 - \frac{-.0268}{1-.0001X}\right]^{\frac{1}{148}}$$
The X = 500 yds

$$\frac{t_2}{t_1} = 1.004 \left(\frac{o_1}{c_9}\right)^{\frac{1}{148}}$$
The 2000 yds

$$\frac{t_2}{t_1} = 1.011 \left(\frac{o_1}{c_9}\right)^{\frac{1}{148}}$$

As the tentative design size of 2.08 ins is close to that of the optimum size of 2.27 ins, it is now more reasonable than before to consider that  $o_1$  and  $o_2$  (usually given as  $\log_{10} c$  in the perforation formula) are constant, and  $W_{23}$  the discard weight, does not vary seriously.

It is also to be noted that the thickness of plate perforated by the optimum size shot of 2.27 diameter compared with that of 2.08 diameter is 0.45 and 1.15 better, that is, for this difference in diameters, the improvement is small.

#### Example 4.

This is the problem of example 2.

As before, the 20 Pr. is taken, but now as the tentative design size. Thus we have

Decimeter of sub-shot = 2.08 ins = d,
Weight of 20 Pr. shot and cap = 5.2 lb = W1
Weight discard = 4.22 lb = W2
Weight of charge = 16.5 lb = C

The valocity for this tentative design is also based on the projectile referred to in example 2 wis. weight 19 lbs with 16.5 lb charge gave a velocity 4400 ft/sec.

So M.V. of the above 20 Pr, projectile is given by

$$(19 + \frac{16.5}{2}) \pm 400^{9} = (5.2 + 4.22 + \frac{16.5}{2}) V_{1}^{9}$$

$$V_{1} = 5464$$

$$b = \frac{16.5}{205.2} + \frac{4.22}{5.2} = 1.586 + .812 = 2.398$$

$$a = 1 + b = 3.398$$

For first approximation  $K_0 = .969 \sqrt[8]{2.398} = 1.297$  from (13. or optimum size = 1.297 x 2.08 = 2.7 ins.

Using eq. 12 to obtain more accurate values, we have

So optimum size for range 500 yds = 1.315 2 2.08 = 2.73 For range 2000 yds = 1.355 x 2.08 = 2.82

or a practical mean of 2.77 which is in good agreement with example 2.

$$K_0$$
 is now  $\frac{2 \cdot 77}{2 \cdot 08} = 1.531$ 

$$\frac{t_n}{t_1} = \left(\frac{c_1}{c_2}\right)^{\frac{1}{1+3}} \frac{1}{1.331^{\circ}}, = \left[1 - \frac{1 - \left(\frac{5.396 \times 1.331^{\circ}}{1.331^{\circ} + 2.396}\right)^{\frac{1}{12}}}{1 - \frac{2.968^{\circ}}{1.656 \times 5.2 \times 54.24}}\right]^{\frac{1}{12}}$$

$$= \left(\frac{c_1}{c_2}\right)^{\frac{1}{14}} \times .872 \left[1 - \frac{-.125}{1 - .0000938}\right]^{\frac{1}{14}}$$

$$\frac{t_2}{t_1} = 1.035 \left(\frac{c_1}{c_2}\right)^{\frac{1}{14}}$$

For I \* 500

$$\frac{t}{t} = 2000 \qquad \frac{t}{t} = 1.065 \left(\frac{c_1}{c_2}\right)^{\frac{1}{1-2}}$$

So at 500 yds the optimus shot is 356 better them the tentative design size of 2.05 ins and at 2000 yds it is 65 better.

#### Second Summary

The results obtained by the alternative method, which relates to the Designers first design, are in agreement with the optimum miss determined in relation to the F.C. size.

It is, however, to be noted that whereas, compared with the R.C. projectile, the ratios  $\xi_0/\xi_0$  decreases with increase of range, the ratio, compared with the Designers tentative size, which in both examples (5) and (4) were smaller than the options sizes determined, increases with increase of range,

#### The A.P.D.S. Projectile - Optimum Sime.

As for the A.P.C.B.C./D.S. projectile, the first object in the following is to determine the optimum size related to the F.C. size, then to obtain the equations related to the Designer's tentative design. This type of D.S. projectile comprises a shot of dense material, a ballistic cap, a sheath which holds shot and cap together, and a discard.

#### Relationship to F.C. Size.

Typical assumblies of the F.C. and D.S. projectile are shown by Figs. 2. Following the previous method we have:-

#### Full Calibra Projectile

#### Discarding Sabot Projectile

Core weight	= V1 = k1d1	Core weight	$= Y_7 = k_1 d_2$
Cap weight	$= \Psi_0 = k_0 \Psi_1 = k_0 k_1 d_1^*$	Cap weight	m Wamkaw m kakida
Sheath weight	= W4 = k <sub>d</sub> W <sub>4</sub> = k <sub>d</sub> k <sub>k</sub> d <sub>4</sub>	Sheath weight	= W <sub>a</sub> = k <sub>a</sub> W <sub>7</sub> = k <sub>a</sub> k <sub>a</sub> da
Charge weight	<b>≡</b> C	Charge weight	₩ Ö Ü
X.V.	<b></b> ∀ <b>y</b>	M.Y.	<b>= T</b> s .
Sheath thickness	$= (D-d_1)/2$	Sheath thickness	$=\frac{(D-d_1)}{2}\cdot\frac{d_2}{d_1}$
Total weight	# W = W1 + W2 + W4	Discard weight	= W
Diameter of F.C.	m D	Dismeter of core	= d <sub>e</sub>
Dismeter of core	= ₫ <sub>1</sub>	Let K	≖ d.√D
Let Z -	= d <sub>1</sub> /D	then da/d1	$=\frac{\mathbf{K}}{2}$

Equating energies:-

#### Range, and Velocity at Target

The target velocity for the F.C. and D.S. Projectile are therefore

$$V_{p_{X}} = V_{p} - \frac{XD^{2}}{1.656(\pi_{1} + W_{2} + W_{4})} \qquad V_{g_{X}} = V_{g} - \frac{X\left[d_{n} + \frac{d_{n}}{d_{n}}(D - d_{n})\right]}{1.656(W + W_{n} + W_{4})}$$

$$= V_{p} - \frac{X}{1.656 \ Z \ Dk_{1}W_{1}} \qquad = V_{p} \left(\frac{a}{X^{2} + c}\right)^{\frac{1}{2}} - \frac{X}{1.656 \ Z^{2} \ k_{1} \ D \cdot \frac{W}{W_{2}} \cdot K}$$

#### Thickness of Plate Perforated

$$t_{p} = \sqrt{\frac{(\mathbf{x}_{1} + \mathbf{x}_{2}) \nabla_{\mathbf{x}_{1}}^{2} \cos^{2} \theta}{C_{p} d_{1}^{2+87}}} \qquad t_{8} = \sqrt{\frac{\mathbf{x}_{1} + \mathbf{x}_{2} \cdot \nabla_{\mathbf{x}_{1}}^{2} \cos^{2} \theta}{C_{S} d_{2}^{2+87}}}$$
and
$$t_{S} = \sqrt{\frac{\mathbf{x}_{1} + \mathbf{x}_{2} \cdot \nabla_{\mathbf{x}_{1}}^{2} \cdot \nabla_{\mathbf{x}_{1}}^{2}$$

and 
$$\frac{\mathbf{t}_{S}}{\mathbf{t}_{\mathbf{y}}} = \sqrt{\frac{\mathbf{s}_{7} + \mathbf{s}_{6}}{\mathbf{s}_{1} + \mathbf{s}_{2}}} \left( \frac{\mathbf{v}_{SX}}{\mathbf{v}_{PX}} \right) \frac{\mathbf{c}_{\mathbf{y}}}{\mathbf{c}_{\mathbf{S}}} \left( \frac{\mathbf{d}_{4}}{\mathbf{d}_{2}} \right)^{\frac{1}{1657}}$$

$$= \sqrt{\frac{\mathbf{d}_{2}}{\mathbf{d}_{3}}} \left( \frac{\mathbf{d}_{4}}{\mathbf{d}_{3}} \right)^{\frac{1}{1657}} \cdot \frac{\mathbf{c}_{\mathbf{y}}}{\mathbf{c}_{\mathbf{S}}} \cdot \frac{1}{f} \left[ \mathbf{v}_{\mathbf{y}} \left( \frac{\mathbf{s}}{\mathbf{K}^{2} + \mathbf{s}_{0}} \right)^{-\frac{1}{2}} - \frac{\mathbf{x}_{\mathbf{w}}}{1.656} \frac{\mathbf{z}_{\mathbf{k}_{1}} \mathbf{D} \mathbf{w}}{\mathbf{x}_{1}} \right]^{\frac{1}{1656}}$$

$$= \left( \frac{\mathbf{c}_{\mathbf{y}}}{\mathbf{c}_{*}} \right)^{\frac{1}{1656}} \frac{1}{fZ} \left[ \mathbf{v}_{\mathbf{y}} \cdot \mathbf{s}^{\frac{1}{2}} \cdot \frac{\mathbf{x}_{1}^{-71.5}}{(\mathbf{x}^{2} + \mathbf{s}_{0})^{\frac{1}{2}} - \frac{\mathbf{h}}{\mathbf{x}} \mathbf{s} \mathbf{s}} \right]^{\frac{1}{1656}}$$

$$(15)$$

which is similar to equation (8

where 
$$f = \left[ V_y - \frac{\lambda V_z}{1.656k, Z^2 VD} \right]^{\frac{1}{2} \frac{1}{4} \frac{1}{2}} = (V_y - \frac{h}{2})$$

$$h = \frac{\chi D^2 Z}{1.656V}$$

Differentiating (19, and equating to zero as before, the optimum size is given by

$$.715 - \frac{3}{2} \frac{K^{a}}{K^{2}+m} + 1 \frac{(K^{a}+m)^{\frac{1}{2}}}{K} = 0$$
 (16)

where

The first approximation to the optimum size is again obtained by neglecting the third value of (16) giving

$$\frac{d_2}{D} = K_0 = .969 \sqrt[8]{n} = .969 \sqrt[8]{\frac{1}{\sqrt{N}} (N_0 + \frac{C}{2})}$$
 (17)

With this as a starting value, the tabular method previously used can be applied to (16) to determine more accurate optimum sizes relating to varying ranges. Then the rates of plate thickness, perforated by this optimum size, to that by the F.C. can be calculated from (15) which reduces to the form

$$\frac{t_{S}}{t_{P}^{2}} = \left(\frac{C_{P}}{C_{S}}\right)^{\frac{1}{2+3}} \cdot \frac{1}{2K} \cdot 10^{4} = \begin{bmatrix} \frac{1}{2} & -\frac{1}{2} & -\frac{1}{(K^{2}+p)^{\frac{1}{2}}} \\ \frac{1}{1-\frac{KD^{2}}{1-\frac{656WV}{2}}} \end{bmatrix}^{\frac{1}{2+3}}$$
(18)

now (17) is the same as (9) with the inclusion of the ratio Z. From current designs the value of Z is of the order .77. The rough rule for A.P.C.B.C./D.S. shot is \$ P.C. bence the rough rule for A.P./D.S. becomes

#### Alternative Solution based on Designers Tentative Design

The designers A.P./D.S. tentative core size can as shown above be taken as § F.C. size. Proceeding as before, we have (see fig. 3)

Tentative Design

W<sub>0</sub> = k<sub>1</sub>d<sub>2</sub>

W<sub>0</sub> = k<sub>2</sub>d<sub>3</sub>

W<sub>10</sub> = k<sub>2</sub>M<sub>0</sub> = k<sub>2</sub>k<sub>1</sub>d<sub>3</sub>

W<sub>10</sub> = k<sub>2</sub>M<sub>0</sub> = k<sub>2</sub>k<sub>1</sub>d<sub>3</sub> Core weight Cap weight Sheath wight

$$K = \frac{d_2}{d_3}$$

Equating Energies we get

$$V_{\underline{A}} = V_{\underline{T}} \left( \frac{1}{\underline{X}^{\underline{A}} + \underline{\alpha}} \right)^{\frac{1}{\underline{A}}}$$

$$a = \frac{1}{W_{T}} \left( \frac{W_{T}}{\underline{X}^{\underline{A}} + \underline{\alpha}} \right)^{\frac{1}{\underline{A}}}$$

$$a = \frac{1}{W_{T}} \left( \frac{W_{T}}{\underline{X}^{\underline{A}} + \underline{\alpha}} \right)^{\frac{1}{\underline{A}}}$$

$$a = \frac{1}{W_{T}} \left( \frac{W_{T}}{\underline{X}^{\underline{A}} + \underline{\alpha}} \right)^{\frac{1}{\underline{A}}}$$

Target Velocities

$$\nabla_{p_{X}} = \nabla_{T} - \frac{X(d_{g} + 2t_{p})^{2}}{1.656(W_{g} + W_{g} + W_{g})} \qquad \nabla_{\underline{A}X} = \nabla_{\underline{A}} - \frac{X(d_{g} + 2t_{p}, \frac{d_{g}}{d_{g}})^{2}}{1.656(W_{g} + W_{g} + W_{10})} \\
= \nabla_{\underline{T}} - p \qquad = \nabla_{\underline{T}} \left(\frac{S}{X + m}\right)^{\frac{1}{2}} - \frac{D}{K}$$

where

$$t_{A} = \sqrt{\frac{1.656(\frac{1}{4} - W_{A})}{\frac{1.48}{2} + \frac{1.48}{2}} \sqrt{\frac{W_{A} + W_{B}}{\frac{1.48}{2}} \sqrt{\frac{W_{A} + W_{B}}{\frac{1.48}{2}}} \sqrt{\frac{W_{A} + W_{B}}{\frac{1.48}{2}} \sqrt{\frac{W_{A} + W_{B}}{\frac{1.48}{2}$$

$$\frac{t_{\underline{A}}}{t_{\underline{T}}} = \sqrt{\frac{W_0 + W_0}{W_7 + W_0}} \left( \frac{\overline{V}_{\underline{A}\underline{X}}}{\overline{V}_{\underline{T}\underline{X}}} \right) \left( \frac{C_{\underline{T}}}{C_{\underline{A}}} \right) \left( \frac{d_{\underline{A}}}{d_{\underline{A}}} \right)^{1-57}$$

$$= \sqrt{\frac{1-4}{C_{\underline{A}}} \frac{\sqrt{C_{\underline{T}}}}{(\overline{V}_{\underline{A}} - D)^2}} \left\{ V_{\underline{T}} = \frac{1}{2} \frac{X^{-7 \pm 8}}{(X^{\frac{5}{2}} + D)^{\frac{1}{2}}} - \frac{D}{X^{-568}} \right\}^{\frac{5}{2} + \frac{1}{2}}$$
(19)

The optimum size is given by

$$.715 - \frac{3}{2} \frac{K^{2}}{(K^{2}+m)} + 1 \frac{(K^{2}+m)^{\frac{1}{2}}}{K} = 0$$
 (20)

apera

and the first approximation by

$$K_0 = .969 \sqrt[4]{n} = .969 \sqrt[4]{\frac{C}{2} + V_0}$$
 (21)

Variation of ratio ta/by with Range

As before, equation (19) reduces to

$$\frac{\xi_{A}}{\xi_{R}} = \begin{pmatrix} Q_{A} \\ C_{A} \end{pmatrix}^{\frac{1}{2}\frac{1}{2}} \frac{1}{K} \exp \left[ 1 - \frac{1 - \left( \frac{R^{2}}{K^{2} + R} \right)^{\frac{1}{2}}}{1 - \frac{2d_{A}}{1 + 656} (\frac{R^{2} - R}{K^{2}}) \frac{R}{R}}} \right]^{\frac{1}{2}\frac{1}{2}}$$
(22)

#### Conclusions

It is shown that rough rules for the optimum size of A.P.C.B.C./D.S. shot and A.P./D.S. cores are 2/3 F.C. and 1/2 F.C. respectively.

& closer value is given by the typical equation

optimum aims ratio K = .969 A \$\sqrt{B}

where A and B relate to functions of dimensions and weights of either the F.C. projectile or Designer's tentative design.

It is also shown that the true optimum size is dependent on the target range and is therefore not constant for all ranges. Nevertheless for practical reasons the size to be selected must be common for all practical ranges and therefore a mean value of the extremes calculated.

The examples show that, though the optimum size shot gives the maximum perforation, the rate of change of thickness perforated, with variation of size of shot around the optimum, is small.

For this reason, and especially when economy in cost and material is necessary it may be considered sufficient to design to a shot size slightly less than the optimum.

#### APPENDIX

An alternative solution to solving the equations  $(B_{\mu}, (1.2, 1.6))$  and (20) all of which are of the form

$$-715 - \frac{3}{2} \frac{K^{2}}{K^{2} + b} + \frac{(K^{2} + b)^{\frac{1}{2}}}{K} = 0$$
 (a)

and having to a first approximation the solution

$$\mathbf{K}_{0} = .969 \sqrt[8]{b} \tag{b}$$

may be obtained by use of the Taylor Theorem. This method can replace the tabular method used in the paper.

The Taylor expansion is given by

$$f(x+h) = f(x) + hf^{1}(x) + \frac{h^{2}}{2}f^{2}(x) = 0$$

taking the first two terms we have

$$p = -\frac{1}{1(x)}$$

From the equations (a) and (b) above we have

$$f(K_0) = .715 - \frac{3}{2} \frac{.969^8 b}{(.969^8 b+b)} + 1 \frac{(.969^8 b+b)^{\frac{1}{2}}}{.969^6} = 1.4262 \cdot 1 b^{\frac{1}{2}}$$

$$f'(X) = -\frac{9}{2} \frac{bX^{0}}{(X^{0}+b)^{2}} + i \left[ \frac{X}{2(X^{0}+b)^{\frac{1}{2}}} - \frac{(X^{0}+b)^{\frac{1}{2}}}{X^{0}} \right]$$

and

$$f'(\mathbf{E}_{b}) = -\frac{9}{2} \frac{bx_{a}}{(.969^{9}b+b)^{3}} + 1 \left[ \frac{3}{2} \frac{.969^{9}b+b}{(.969^{9}b+b)^{\frac{1}{2}}} - \frac{(.969^{9}b+b)^{\frac{1}{2}}}{.969^{9}b^{\frac{1}{2}}} \right]$$
$$= -\left[ \frac{1.158}{b} + .3745 \frac{1}{b} \right]$$

Hence the solution becomes

$$.969b^{\frac{1}{3}} + \frac{1.4262 i b^{\frac{1}{6}}}{b^{\frac{1}{3}} + .3745 \frac{1}{b^{\frac{1}{6}}}}$$

$$= \left[.969 + \frac{1.4262 i b^{\frac{1}{6}}}{1.158 + .3745 i b^{\frac{1}{6}}}\right] b^{\frac{1}{3}}$$

$$= \left[.969 + 1.2516 i b^{\frac{1}{6}}\right] b^{\frac{1}{3}} \text{ with sufficient accuracy} \qquad (o)$$

For example, applying this to example 1 and for the range X = 2000 yds. we had b = .328, t = .0472

From (c) above - Opt. size ratio =  $.3284^{\frac{1}{3}}(.969+1.2316x.0472x.382i^{\frac{1}{6}})$ = .6899(.969+.0483) = .7027

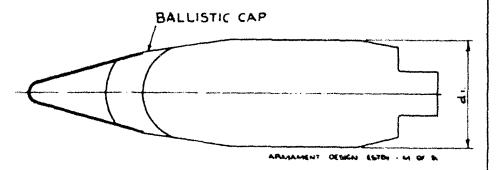
or optimum size =  $3.3 \times .7027 = 2.32$  ins. as before

For example 3 we had b = 1.3115, and = .0372 at 2000 yds range

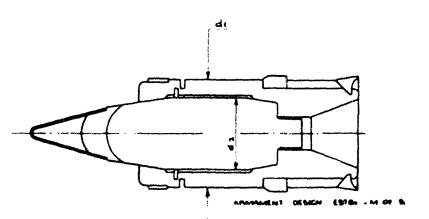
then optimum size ratio = 1.3115<sup>3</sup> (.969+1.2316x.0372x1.3115<sup>3</sup>) = 1.0946(.969+.048) = 1.113

Optimum mime =  $2.08 \times 1.113 = 2.32 \text{ ins.}$  as before.

The state of the s

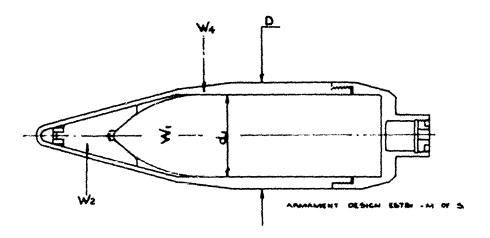


#### APCBC - FULL CALIBRE SHOT.

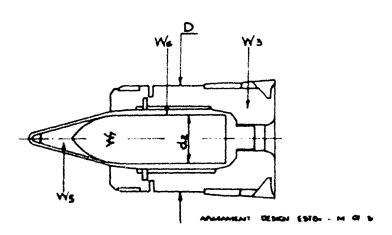


A P.C B.C./D.S PROJECTILE.

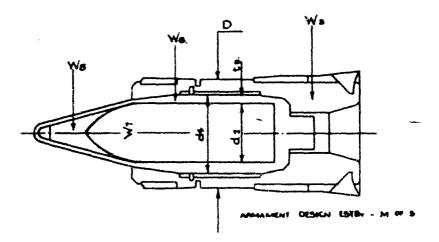
SECRET



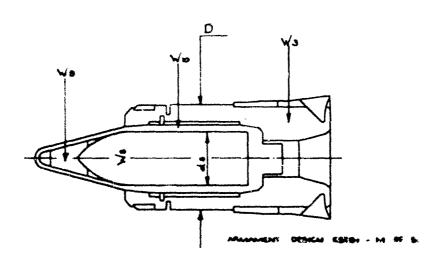
AP - FULL CALIBRE



A.P. O.S. - PROTECTILE



## A P/D.5 TENTATIVE DESIGN.



A P/DS OPTIMUM SIZE.



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